

# Computational Modeling of Brittle Materials Under Dynamic Conditions

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## **Outline**

- **Experimental observations**
- **SCRAM and DCA (Dominant Crack Algorithm) Models:**
  - *Continuum-level* models based on statistical consideration of defects.
- **Calculations and data for explosives & ceramics**
- **Summary/discussions**

**Main Collaborators:**

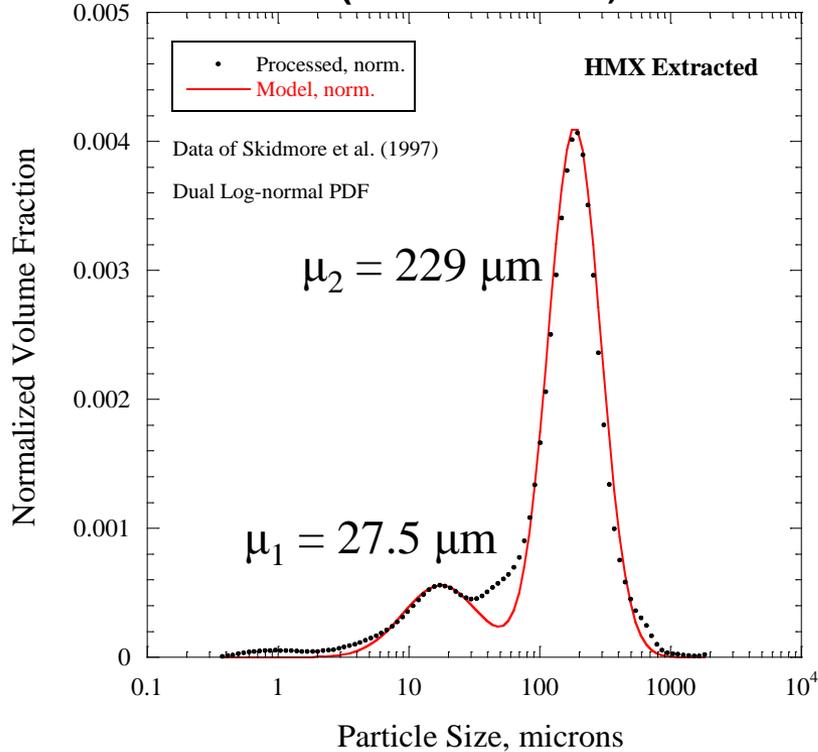
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Wang, Evans, Toutanji (UAH);  
Dienes, Addressio, Lewis (LANL)**

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**Damage and Failure in Brittle Materials**  
**(explosives, ceramics, propellants, concrete, *sea ice*, etc.)**

# Cracking in an Explosive (PBX 9501)

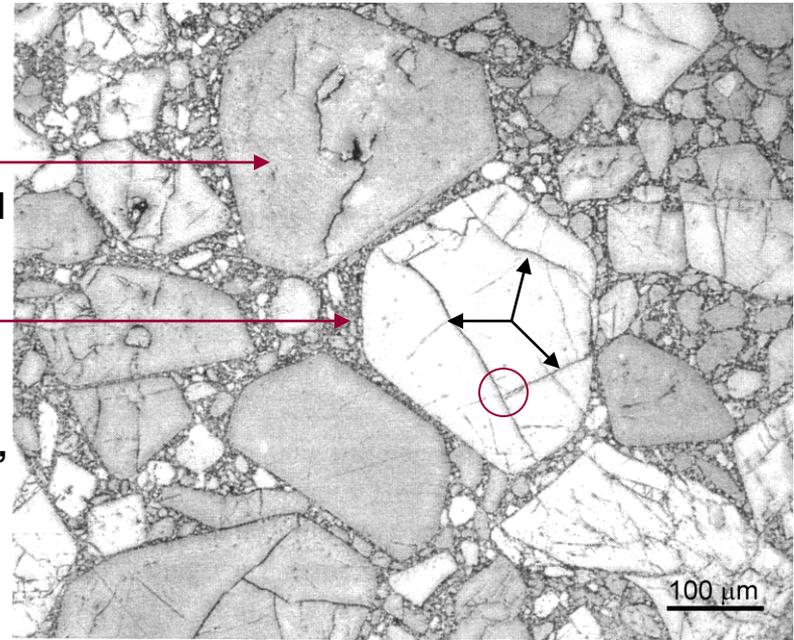
size distribution of HMX Grains  
(Bronkhorst et al.)



HMX  
Crystal

Binder

“Pristine”  
sample



Target 8

Disk of  
explosive  
after impact

Idar et al, 1998

Circumferential  
Cracks

Radial  
Cracks

5"

**Heterogeneous:**

**HMX crystals & Polymer Binder**

Distribution of HMX grains is random:

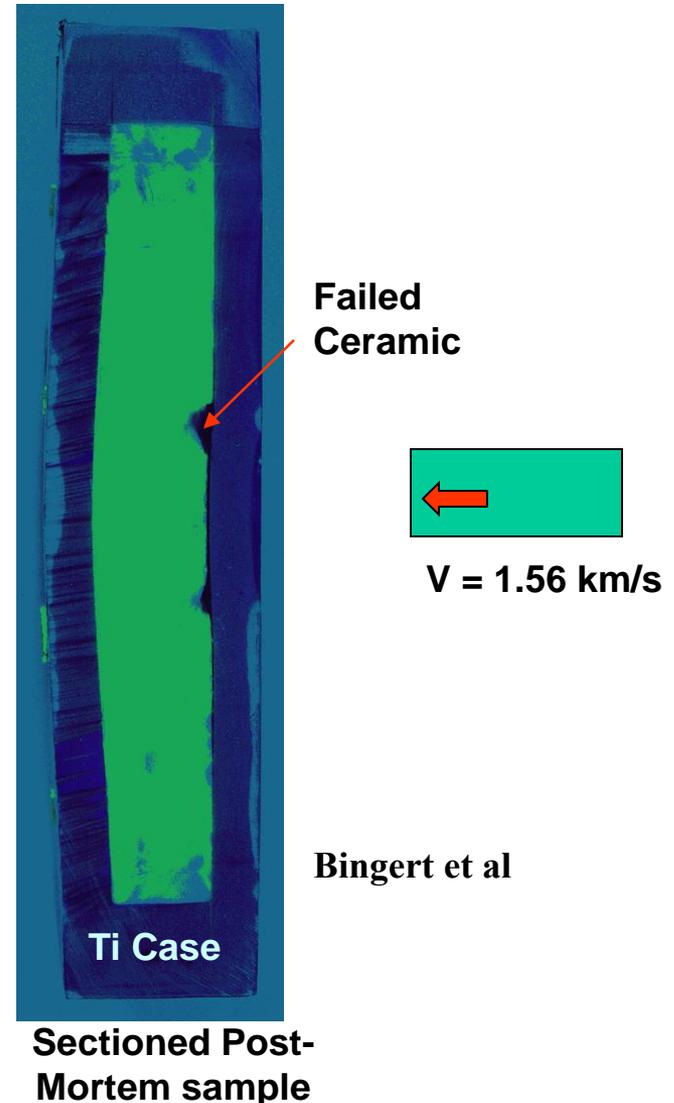
**Bi-modal** distribution (229/27.5  $\mu\text{ms}$ )

**Microcracks** in the “pristine” sample:  
different **sizes & orientations**

# Ceramic Armor under Ballistic Impact

- **Ceramic** disk: Coors AD 995 (1/2" thick, 4 in diameter)
- **Casing (3 pieces):** Ti-6Al-4V  
2 cover plates (1/4 " thick each)  
Ring (5" diameter, 1/2 " thick)
- **Impactor:** Lexan rod  
1.5" long , 3/4 " diameter
- **Impact velocity:** 1.56 km/s

Can we predict the *location and size* of the failed ceramic?



# Computational Modeling

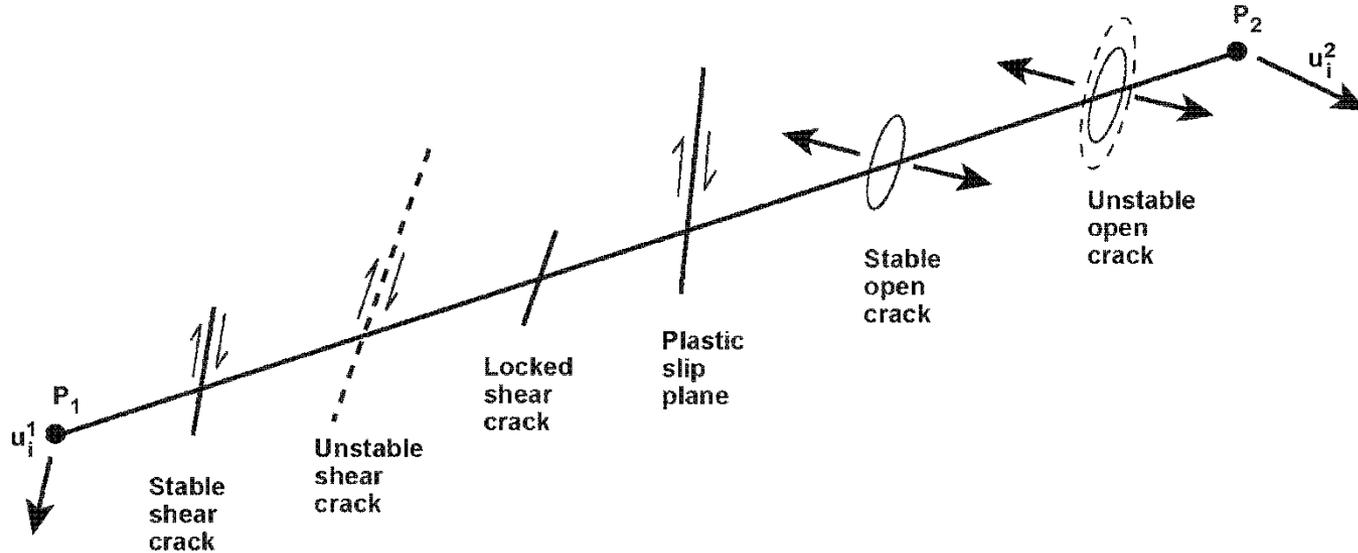
# Challenges

- Events taking place at length scales below the *grid resolution* can have significant effects on the damage and failure
- Challenges:
  - how far do we need to go down in *scales* (meso/micro scales?)
  - What about the history of *defects*?
  - How do we connect the mechanics/physics of different scales? (e.g., grain-level mechanics)
  - What about convergence and stability of the *numerical solution* (i.e., *ill-posedness*/Hadamard instability)?

## Models based on statistical consideration of defects (cracks)

- **SCRAM**
- **DCA**

# Superposition of Strain Rates



## Defects idealized in the model

Difference in velocity

$$\Delta \mathbf{u} \equiv \mathbf{u}^2 - \mathbf{u}^1 = \sum \Delta \mathbf{u}^c + \sum \Delta \mathbf{u}^d$$

Discont.

Continuous Deformation

Statistical averaging  
(probability density function)



$$\mathbf{d} = \mathbf{d}_m + \mathbf{d}_c + \mathbf{d}_g + \mathbf{d}_p$$

Total Stretch (rate of deformation)  $\mathbf{d} \equiv (\nabla \mathbf{u})_{sym}$

Matrix  $\mathbf{d}_m$

Crack  $\mathbf{d}_c$

Plastic  $\mathbf{d}_p$

Crack growth  $\mathbf{d}_g$

**SCRAM: Model for damage, failure and  
initiation of brittle materials**

# Impact Initiation Scenario

Shock wave activates shear cracks.



**Shock  
Compression**

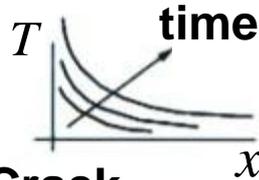


Shear cracks grow easily in HMX,  
inhibited by binder.

Shear cracks grind and produce heat.



$x$  **Normal to  
Shear crack**



Heat conducted away from sliding interface; temperature rise causes cooking of HMX.  $\dot{T} = DT_{xx} + qZe^{-E/kT}$

**Sub-grid model**

**Crack**

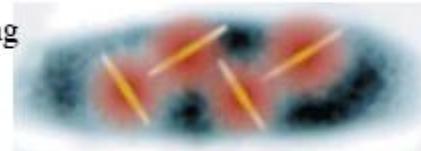
HMX reaches ignition point (Linan & Williams or IGNC)  $T > T_c$



Burned gases open cracks, create initial high-pressure zone. Opening inhibited by inertia, stiffness, and damping.

$$m\ddot{w} + c\dot{w} + kw = pA \quad p = f(M, T)$$

Burning increases general pressure, accelerating burn in adjoining hot spots.

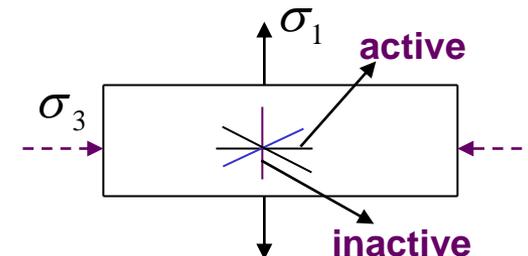


Cracks coalesce when percolation threshold is exceeded, causing general explosion.

# Dominant Crack Algorithm (DCA) Model (Zuo et al., 2006- )

- **Damage surface** is derived from the instability condition for the **critical crack orientation**;
- New damage surface has similar features to that in ISOSCM, but **removes a discontinuity** existed in ISOSCM;
- **Crack opening** strain is **more consistent** with physics-- only the **tensile** principal stresses **contribute** to the crack **opening** strain. Material response can be anisotropic.
- Damage evolution (**growth rate** of the **mean crack radius**) is given by the energy release rate for the crack along the **critical** (most unstable) orientation.
- Including a **nonlinear EOS** and porosity growth.

Zuo, et al., (2006, 2008, 2010, 2011, 2012)



# Crack strain (damage)

Total strain is the sum of **matrix strain** and **crack strain**:

$$\boldsymbol{\varepsilon} = \boldsymbol{\varepsilon}_m(\boldsymbol{\sigma}) + \boldsymbol{\varepsilon}_c(\boldsymbol{\sigma}, \bar{c})$$

**Matrix strain:**

$$\boldsymbol{\varepsilon}_m(\boldsymbol{\sigma}) = \mathbf{C}^m \boldsymbol{\sigma} \quad \mathbf{C}^m = \text{Matrix Compliance}$$

**Crack strain:**

$$\boldsymbol{\varepsilon}_c(\boldsymbol{\sigma}, \bar{c}) = \sum_{\Omega, c} \Delta \boldsymbol{\varepsilon}_c(\boldsymbol{\sigma}, c, \mathbf{n})$$

**For a crack set:**

$$\Delta \boldsymbol{\varepsilon}_c(\boldsymbol{\sigma}, c, \mathbf{n}) = \beta c^3 \left( \mathbf{c}^o + \mathbf{c}^s \right) \boldsymbol{\sigma} n(c, \mathbf{n}, t) \Delta c \Delta \Omega$$

$n(c, \mathbf{n}, t)$ : **crack number density dist.**

**Fabric tensors :**

$$\mathbf{c}^o = \mathbf{b} - \nu \mathbf{a} \quad \mathbf{c}^s = \mathbf{b} - 2\mathbf{a}$$

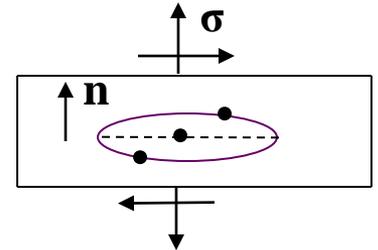
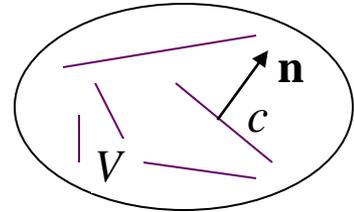
$$\mathbf{a} = \mathbf{n} \otimes \mathbf{n} \otimes \mathbf{n} \otimes \mathbf{n} \quad \mathbf{b} = \mathbf{i} \wedge \mathbf{n} \otimes \mathbf{n} + \mathbf{n} \otimes \mathbf{n} \wedge \mathbf{i}$$

**Material damage:**

$$\boldsymbol{\varepsilon}_c(\boldsymbol{\sigma}, \bar{c}) = \mathbf{D} \boldsymbol{\sigma} = \left( \mathbf{C}^o + \mathbf{C}^s \right) \boldsymbol{\sigma}$$

$$\mathbf{C}^o = (2 - \nu) \beta \int_{\Omega} \int_c n(c, \mathbf{n}, t) \mathbf{c}^o c^3 dc d\Omega$$

$$\mathbf{C}^s = \beta \int_{\Omega} \int_c n(c, \mathbf{n}, t) \mathbf{c}^s c^3 dc d\Omega$$



added compliance

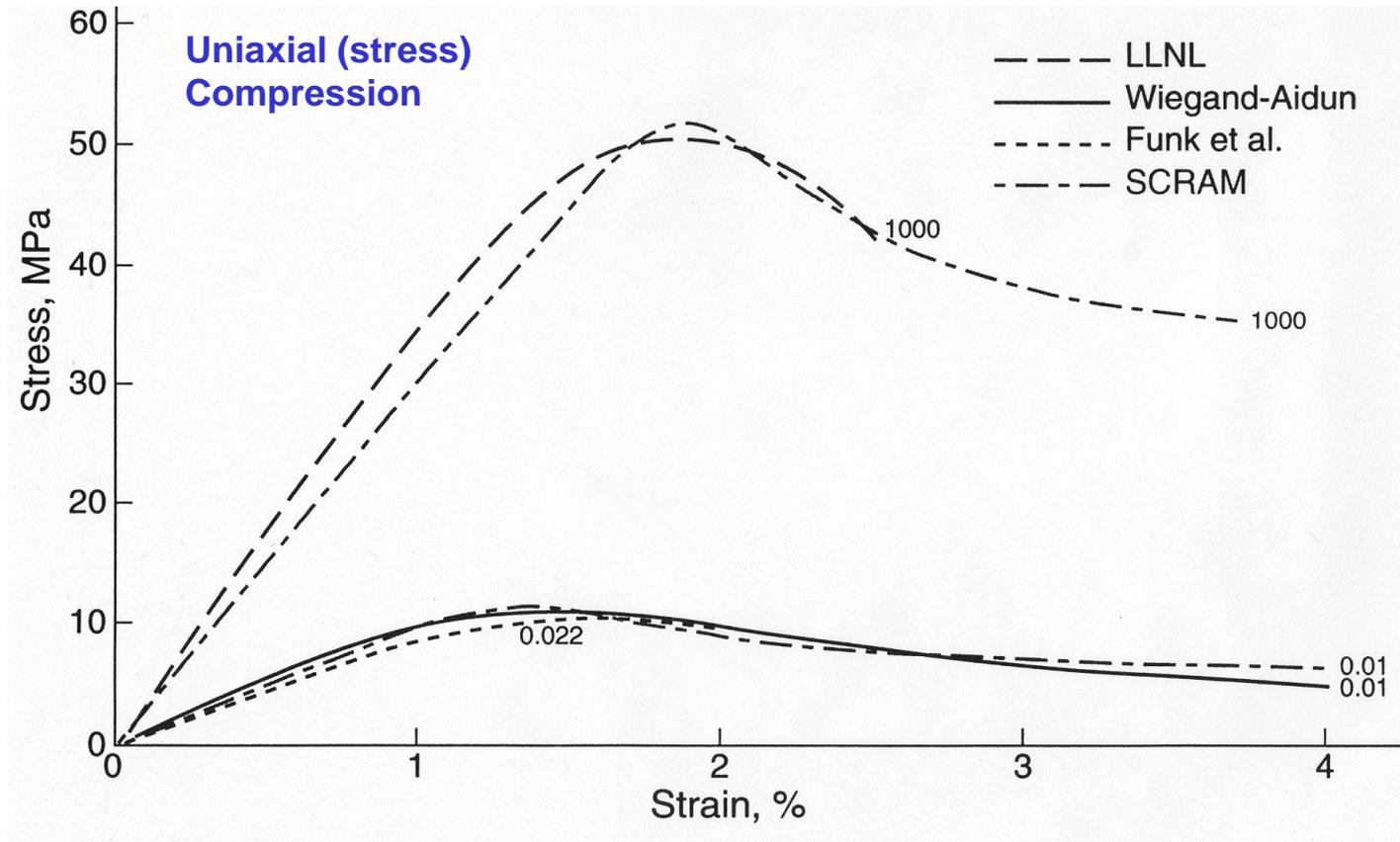
open cracks

shear cracks

**Model calculations and data  
(explosives & ceramics)**

# Comparison of the Stress-Strain Responses with Data

PBX (Plastic- Bonded eXplosive) 9501  
at two strain rates (0.01/s and 1000/s)



**Model captures the data, including peak stress, failure strain, and strain softening**

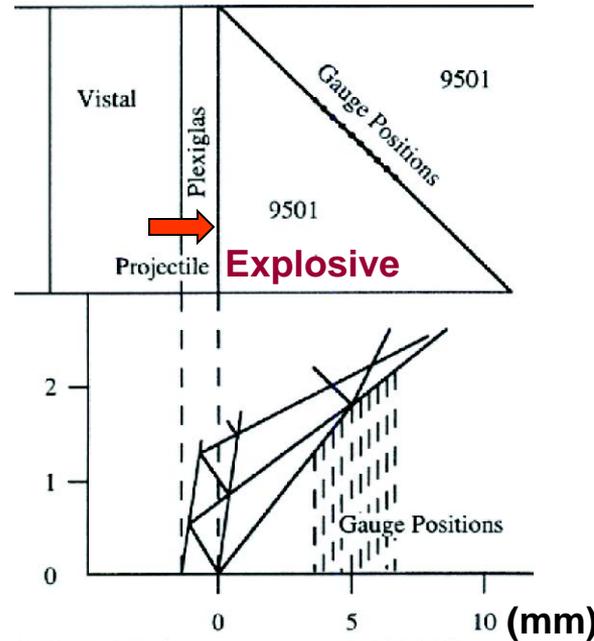
# Comparison with the Multiple-shock Experiment

- Experiment by Mulford et al. of LANL

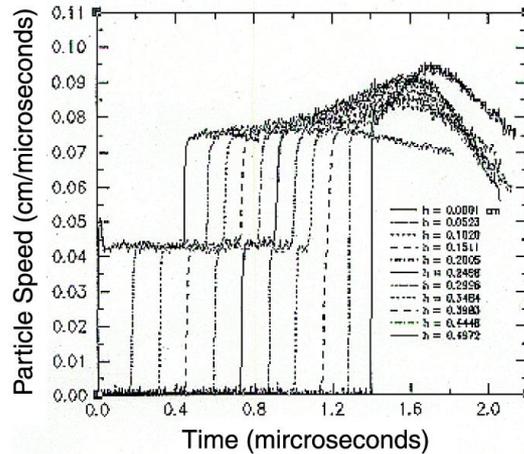
- Composite Impactor: Vistal ( $\text{Al}_2\text{O}_3$ ) and Plexiglas (Kel-F);

- Plate thicknesses (mm):  
11 (Vistal) , 0.8 (Kel-F), 10 (HE);

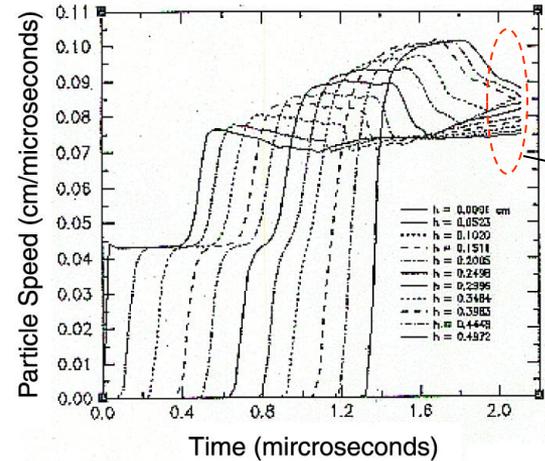
- Impact velocity: 911 m/s



Calculated particle velocity histories agree with data reasonably well.



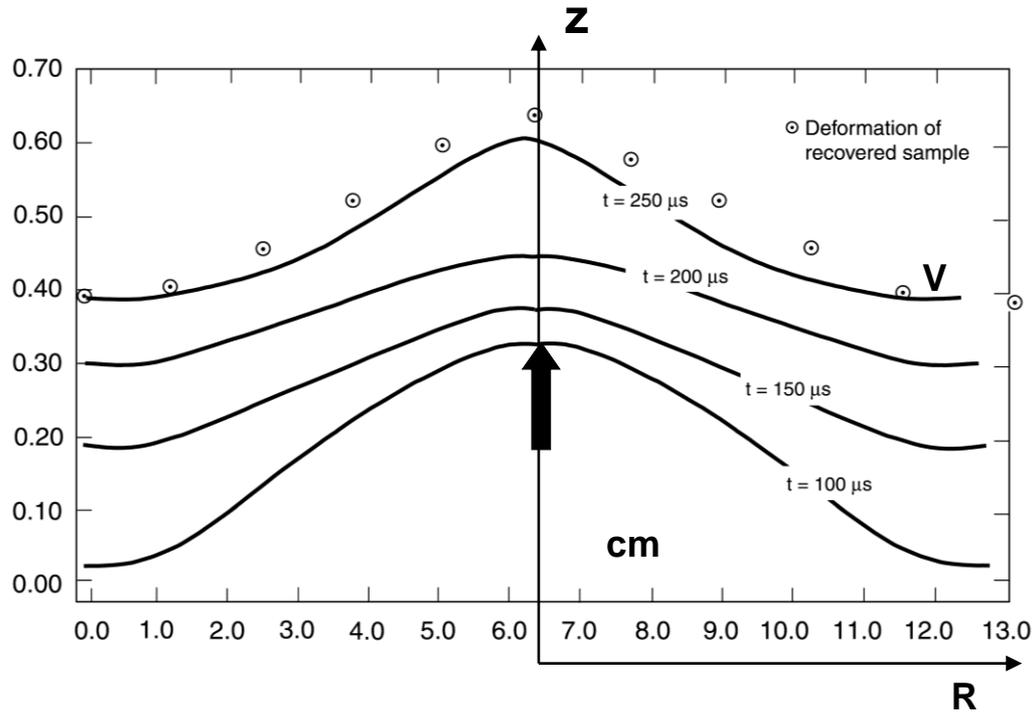
Measured



Calculated

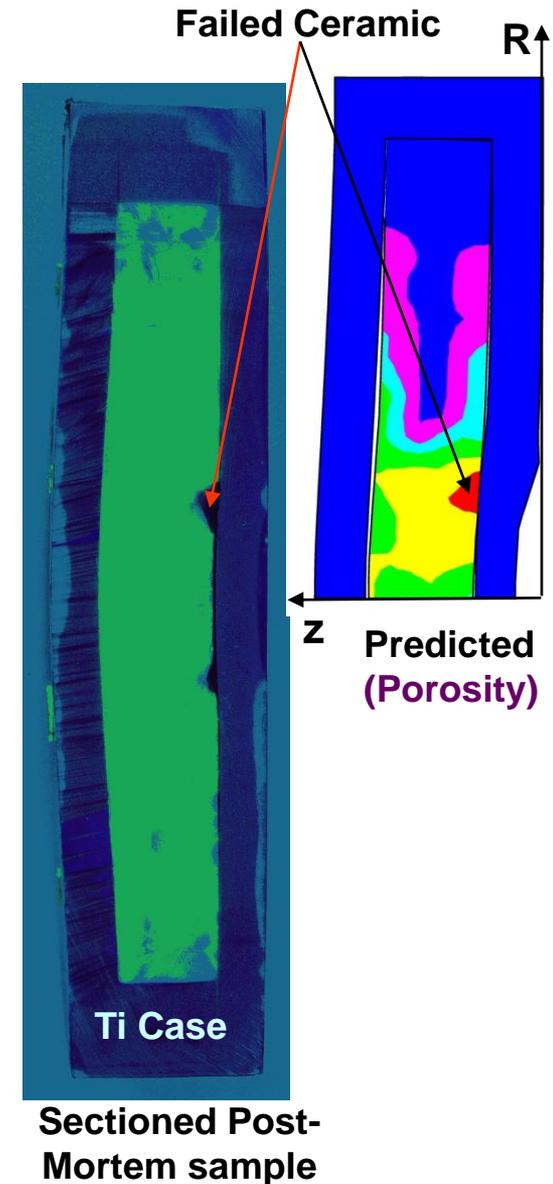
# Modeling Damage and Failure of Ceramic Armor

## Comparison

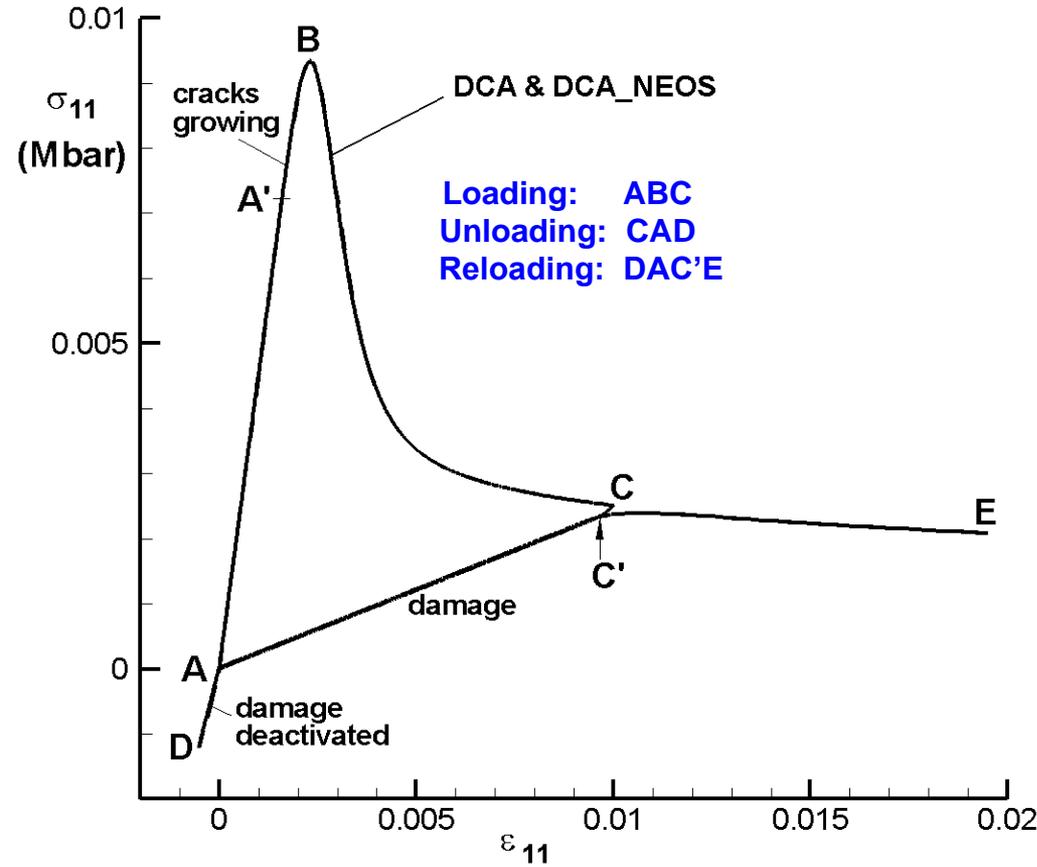


- Model calculation matches the **deformed profile**

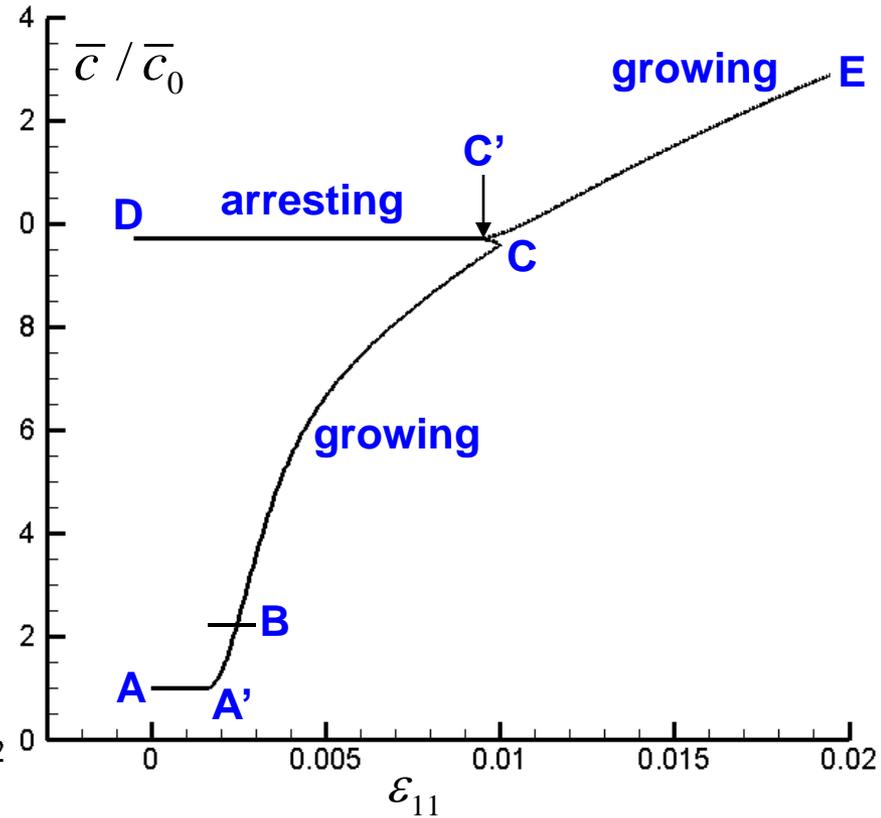
- Calculated location of **failure** (maximum porosity) agrees with the post-mortem observation



# Uniaxial Strain – Cyclic loading Tension/Compression-I

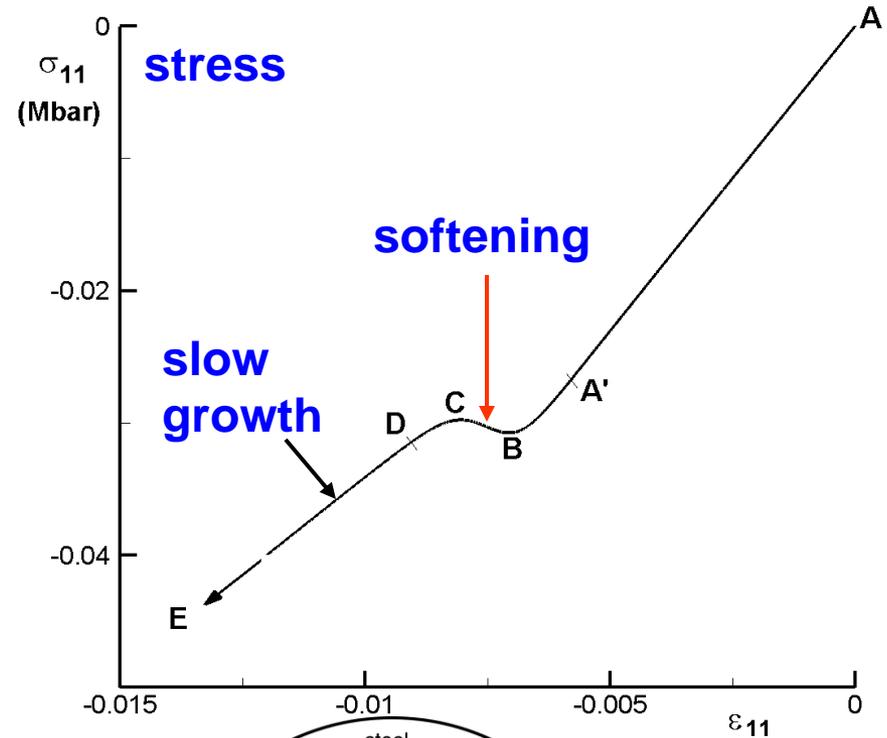
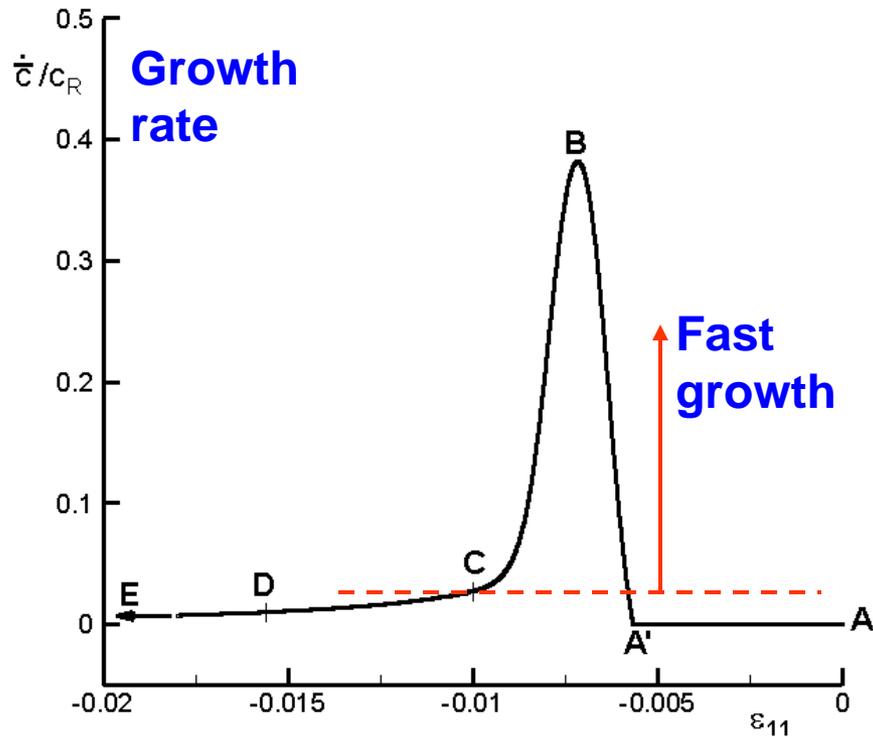


**Stress-strain**

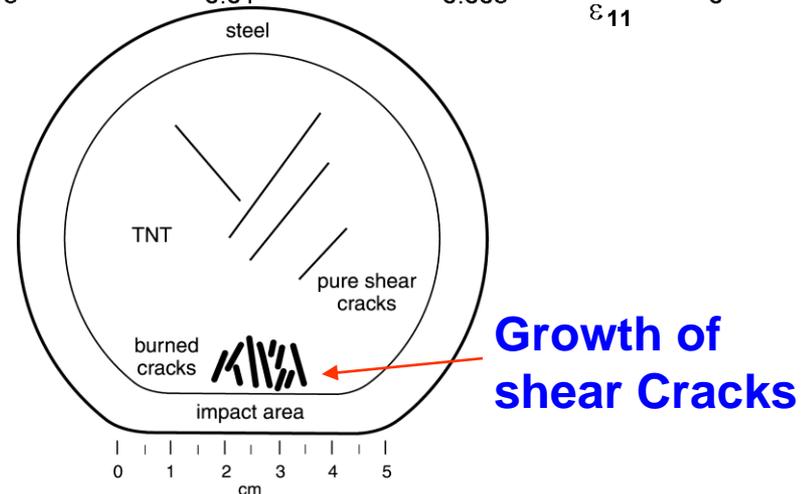


**Evolution of crack size**

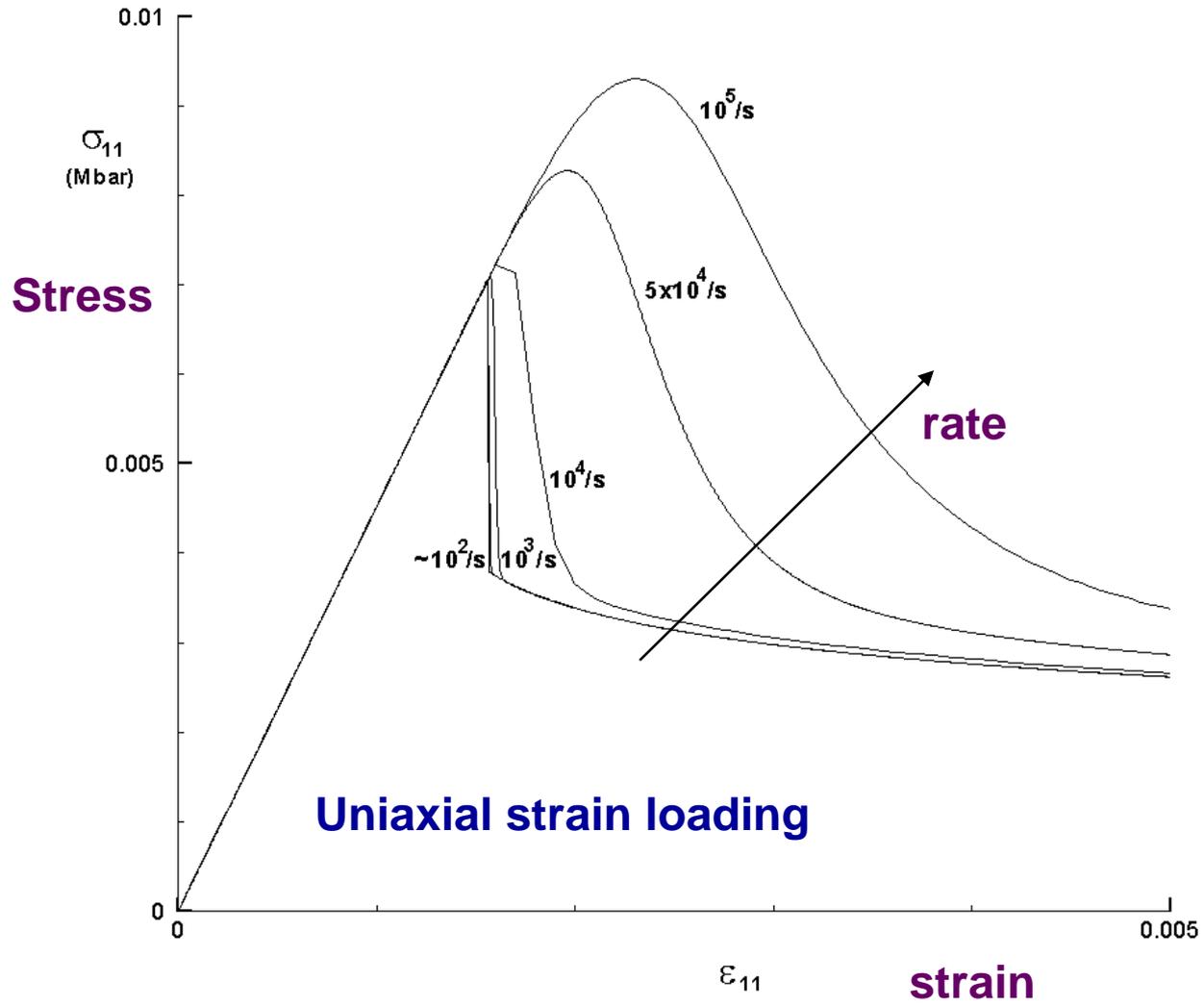
# Crack growth under *Uniaxial* strain compression



- Cracks becomes *unstable* when shear overcomes friction;
- Damage accumulates over a period;
- Stabilized as friction (pressure) takes over



# Rate Effects



# Summary/Discussions

- **SCRAM**: three-dimensional framework for anisotropic damage & failure of *brittle* materials. It also models the *initiation* of chemical reactions (*explosion*) of energetic materials.
- **DCA**: based on SCRAM model but is significantly simpler. It emphasizes on the growth of dominate (critical) crack. No chemical reactions.
- Many **challenges** exists both in fundamental understanding of the response of brittle materials under high rate conditions and in the representation in analysis (*continuum-level*) codes.

# **Extra: Supporting Modeling**

# ISO-SCM Model (Addessio-Johnson)

- A **continuum damage model** based on Dienes' SCM work.
- Key assumptions/limitations: distribution of cracks remain **isotropic**. Damage is also **isotropic**.
- The size distribution remains **exponential** (Seaman et al, SRI), with the mean crack size (damage) evolving with loading.
- Simple **damage surface** based on **averaging** the instability conditions over **all crack orientations**. Damage surface is discontinuous at  $p=0$ .
- The crack strain is a **simple** function of stress and mean crack size. Crack opening strain accounted for  $p<0$  (tension) only.

# Crack Shear and Opening

- **Exponential crack size (radius) distribution (Seaman et al, SRI):**

$$n(c, \mathbf{n}, t) = \frac{N_0(\mathbf{n})}{\bar{c}(\mathbf{n}, t)} \exp\left(-\frac{c}{\bar{c}(\mathbf{n}, t)}\right)$$

$N_0(\mathbf{n}) = N_0$  **Initial crack number density (# per unit vol.); isotropic**

$\bar{c}(\mathbf{n}, t)$ : **Average** radius of the cracks with normal ( $\mathbf{n}$ )

- **SCRAM** accounts for material **anisotropy** by allowing **each direction** has its **own** mean size:  $\bar{c}(\mathbf{n}, t)$  depends on the crack normal ( $\mathbf{n}$ )

- **ISO-SCM (Visco-Scram, etc):** the cracks remain **isotropic**:

$$\bar{c}(\mathbf{n}, t) = \bar{c}$$

**Crack shear strain:**

$$\boldsymbol{\varepsilon}_c^s(\boldsymbol{\sigma}, \bar{c}) = \beta_1 d(\bar{c}) \boldsymbol{\sigma}^d$$

For any  $p$

**Crack opening strain:**  $\boldsymbol{\varepsilon}_c^o(\boldsymbol{\sigma}, \bar{c}) = (2 - \nu) \beta_1 d(\bar{c}) \left( \boldsymbol{\sigma} - \frac{3}{2} p \mathbf{i} \right)$  For  $p < 0$

$$\boldsymbol{\varepsilon}_c^o(\boldsymbol{\sigma}, \bar{c}) = 0$$

For  $p \geq 0$

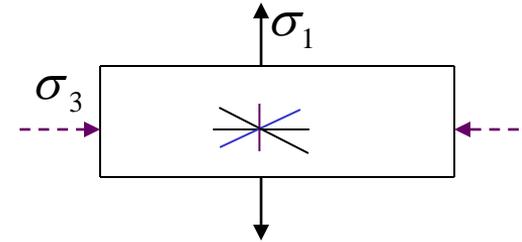
$$\beta_1 = \frac{64}{5G} \frac{1-\nu}{2-\nu}; \quad d(\bar{c}) \equiv N_0 \bar{c}^3 \text{ -damage}$$

# Dominant Crack Model (DCA)

## - Crack Opening strain

Mixed stress states:

$$\sigma_1 > 0; \sigma_3 < 0$$



Only the directions with  $\sigma_n \equiv \mathbf{n} \cdot \boldsymbol{\sigma} \mathbf{n} > 0$  should contribute

ISO-SCM:

$$p = -\frac{\sigma_1 + \sigma_2 + \sigma_3}{3}$$

$p < 0$ , **All directions contribute**

$p \geq 0$ , **No direction contributes**

$$\boldsymbol{\varepsilon}_c^o(\boldsymbol{\sigma}, \bar{c}) = 2\beta_1 D(\bar{c}) \left( \boldsymbol{\sigma} + \frac{1}{2} \text{tr}(\boldsymbol{\sigma}) \mathbf{i} \right)$$

$$\boldsymbol{\varepsilon}_c^o(\boldsymbol{\sigma}, \bar{c}) = \mathbf{0}$$

“Improved”:

$$\boldsymbol{\varepsilon}_c^o(\boldsymbol{\sigma}, \bar{c}) = 2\beta_1 D(\bar{c}) \mathbf{P}^+ \left( \mathbf{P}^d + \frac{5}{2} \mathbf{P}^{sp} \right) \mathbf{P}^+ \boldsymbol{\sigma}$$

Projection operators:

$$\mathbf{P}^{sp} = \frac{1}{3} \mathbf{i} \otimes \mathbf{i} \quad \mathbf{P}^d = \mathbf{I} - \mathbf{P}^{sp} \quad \mathbf{P}^+ \equiv \mathbf{Q}^+ \wedge \mathbf{Q}^+$$

Positive spectral tensor:

$$\mathbf{Q}^+ \equiv \sum_{i=1,3} H[\sigma_i] \mathbf{e}_i \otimes \mathbf{e}_i$$

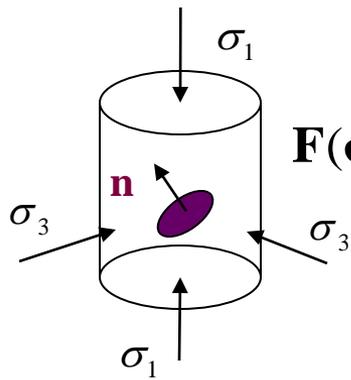
$$\sigma_1 > \sigma_3 > 0$$

$$0 > \sigma_1 > \sigma_3$$

⇒ **ISO-SCM!**

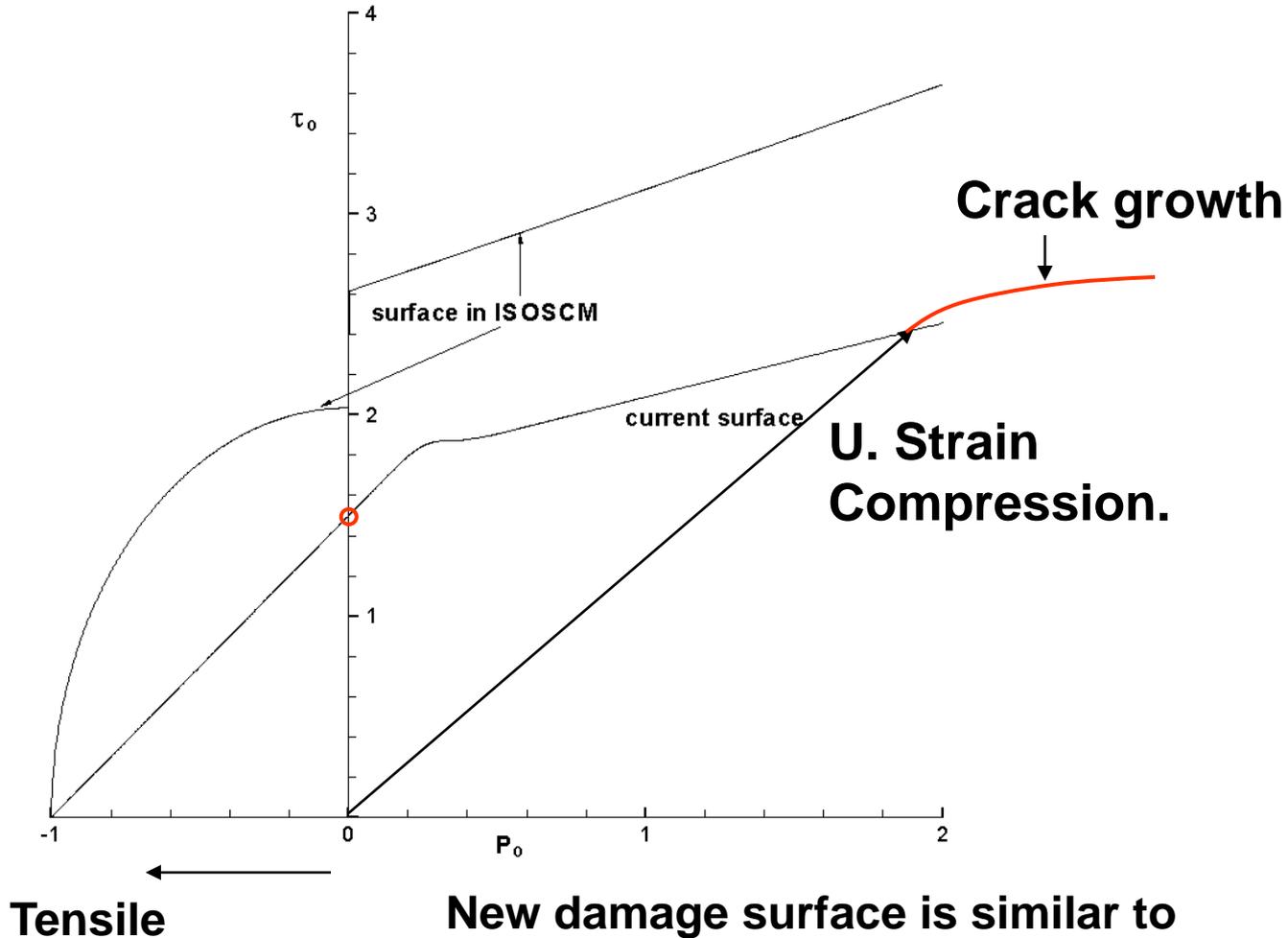
**Only tensile principal stresses contribute to open strain**

# Comparison of the Damage Surfaces



$$\mathbf{F}(\boldsymbol{\sigma}, \bar{\mathbf{c}}) \equiv g(\boldsymbol{\sigma}, \mathbf{n}^c, \bar{\mathbf{c}}) / 2\gamma - 1 \Rightarrow$$

**Dam. Surf.  $\mathbf{F}(\boldsymbol{\sigma}, \bar{\mathbf{c}}) = 0$**



**New damage surface is similar to that in ISO-SCM, but it is continuous.**

## Evolution equation for damage (Crack growth rate)

$$\text{Dam. Surf. } F(\boldsymbol{\sigma}, \bar{c}) = 0$$

Damage increases for stress state **outside** the surface ( $F(\boldsymbol{\sigma}, \bar{c}) > 0$ ):

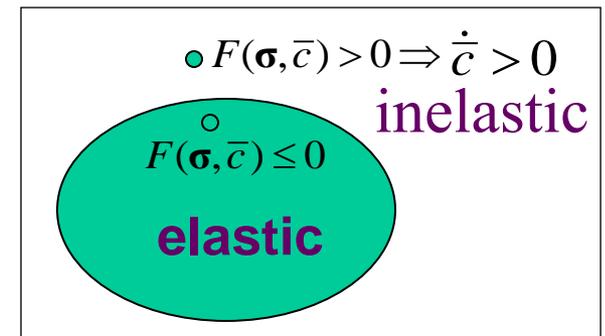
Dynamic crack growth (Freund, 1990):

$$\dot{\bar{c}} = \dot{c}_{\max} \left( 1 - \frac{2\gamma}{g(\boldsymbol{\sigma}, \mathbf{n}^c, \bar{c})} \right) \dot{c}_{\max} : \text{Terminal growth speed (Rayleigh wave speed)}$$

Damage function is also based on  $g(\boldsymbol{\sigma}, \mathbf{n}^c, \bar{c})$ :  $\left( F(\boldsymbol{\sigma}, \bar{c}) \equiv \frac{g(\boldsymbol{\sigma}, \mathbf{n}^c, \bar{c})}{2\gamma} - 1 \right)$

Crack growth is related to the damage function:

$$\dot{\bar{c}} = \dot{c}_{\max} \left( 1 - \frac{1}{1 + \langle F(\boldsymbol{\sigma}, \bar{c}) \rangle} \right)$$



# Stress-strain Relationship

Given a total (applied) strain rate, we need to update stress and damage.

Recall:

$$\boldsymbol{\varepsilon} = \left( \mathbf{C}^m + \mathbf{D}(\bar{c}) \right) \boldsymbol{\sigma}$$

Damage: 
$$\mathbf{D}(\bar{c}) = \beta^e N_0 \bar{c}^3 \left( \frac{3}{2-\nu} \mathbf{P}^d + \mathbf{P}^+ \left( \mathbf{P}^d + \frac{5}{2} \mathbf{P}^{sp} \right) \mathbf{P}^+ \right)$$

Rate form:

$$\dot{\boldsymbol{\varepsilon}} = \mathbf{C}^m \dot{\boldsymbol{\sigma}} + \left[ \mathbf{D}(\bar{c}) \dot{\boldsymbol{\sigma}} + \dot{\mathbf{D}}(\bar{c}) \boldsymbol{\sigma} \right] = \mathbf{C}^m \dot{\boldsymbol{\sigma}} + \left[ \dot{\boldsymbol{\varepsilon}}^c + \dot{\boldsymbol{\varepsilon}}^{gr} \right] \quad (1)$$

strain rate due to crack growth:

Stationary      Crack growth

$$\dot{\boldsymbol{\varepsilon}}_c^{gr} = \dot{\mathbf{D}}(\bar{c}) \boldsymbol{\sigma} = \left( 3\dot{\bar{c}} / \bar{c} \right) \mathbf{D}(\bar{c}) \boldsymbol{\sigma}$$

Stress rate:

$$\dot{\boldsymbol{\sigma}} = -\dot{P}\mathbf{i} + \dot{\mathbf{s}}$$

Deviatoric rate:

$$\mathbf{P}^d \text{Eq.(1)} \Rightarrow \frac{1}{2G_s} \dot{\mathbf{s}} + \mathbf{P}^d \mathbf{D}(\bar{c}) \left( \dot{\mathbf{s}} - \dot{P}\mathbf{i} \right) + \dot{\bar{c}} \frac{3}{\bar{c}} \mathbf{P}^d \mathbf{D}(\bar{c}) \left( \mathbf{s} - P\mathbf{i} \right) = \dot{\boldsymbol{\varepsilon}} \quad (2a)$$

$\dot{P}?$

# Volumetric Response

Pressure in the porous (voided) material

$$P(\phi, \rho, e) = (1 - \phi)P_s(\rho_s, e_s)$$

Mie-Gruneisen equation of state  
(Addressio and Johnson, 1990)

The porosity:

$$\phi \equiv \frac{V_v}{V_s + V_v}$$

Pressure in the solid:  $P_s(\rho_s, e_s) = P_H(\mu_s) \left( 1 - \frac{1}{2} \Gamma_s \mu_s \right) + \Gamma_s \rho_s e_s$

Hugoniot Pressure:

$$P_H(\mu_s) = c_v \mu_s + d_v \mu_s^2 + s_v \mu_s^3$$

Compression ratio:

$$\mu_s \equiv \frac{\rho_s}{\rho_{s0}} - 1$$

Nonlinear effect

Porosity evolution:

$$\dot{\phi} = (1 - \phi) \left( \dot{\epsilon}_{kk}^c + \dot{\epsilon}_{kk}^{gr} \right)$$

TEPLA model:

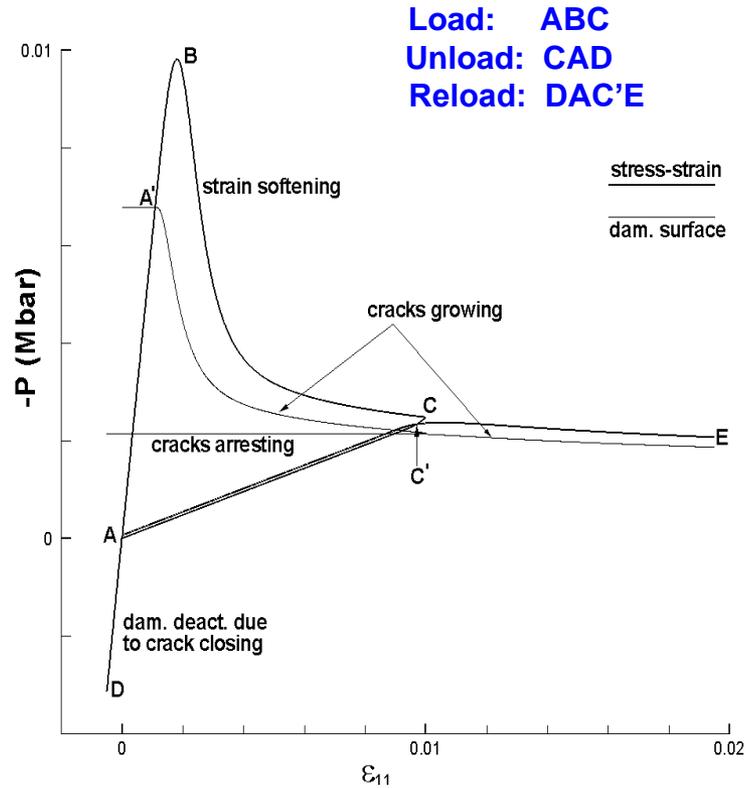
$$\dot{\phi} = (1 - \phi) \dot{\epsilon}_{kk}^p$$

Rate form:

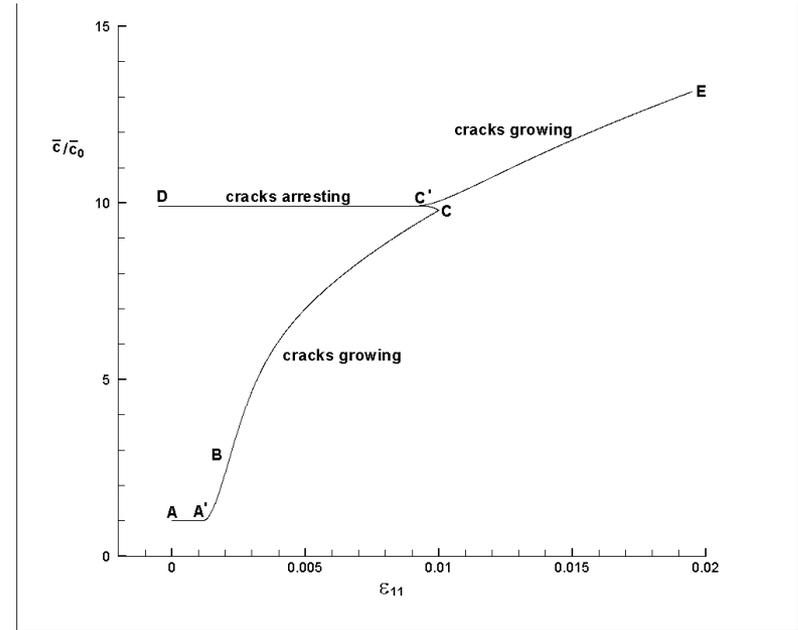
$$\dot{P}(\phi, \rho, e) = -B \dot{\epsilon}_{kk} + \alpha \left( \dot{\epsilon}_{kk}^c + \dot{\epsilon}_{kk}^{gr} \right) + \Gamma_s \mathbf{s} : \dot{\mathbf{e}} \quad (2b)$$

stationary

# Vol. Tension/Compression

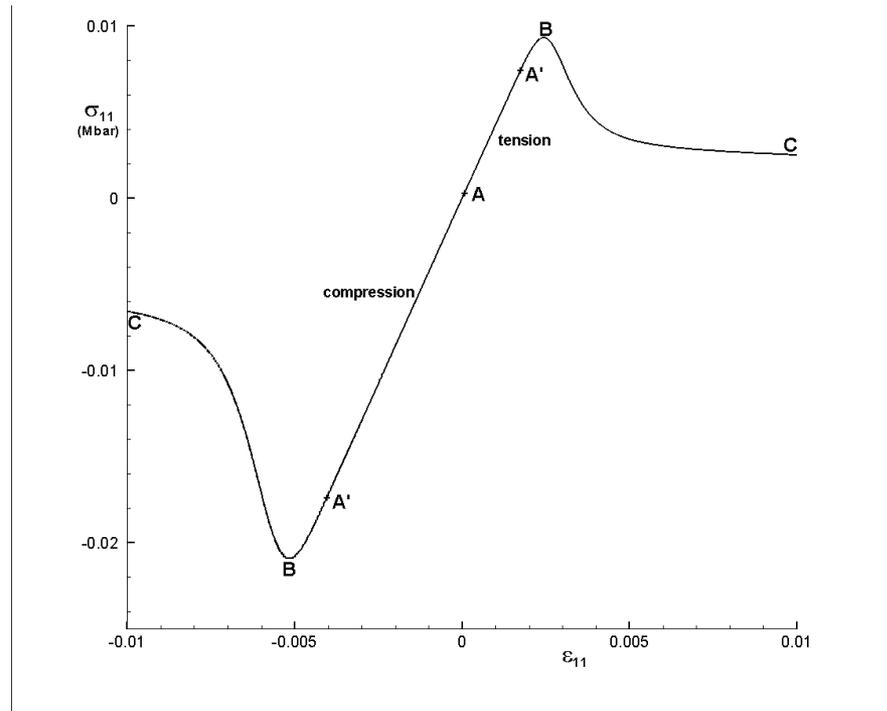


Pressure-strain



Evolution of damage

# Uniaxial (stress) tension and compression



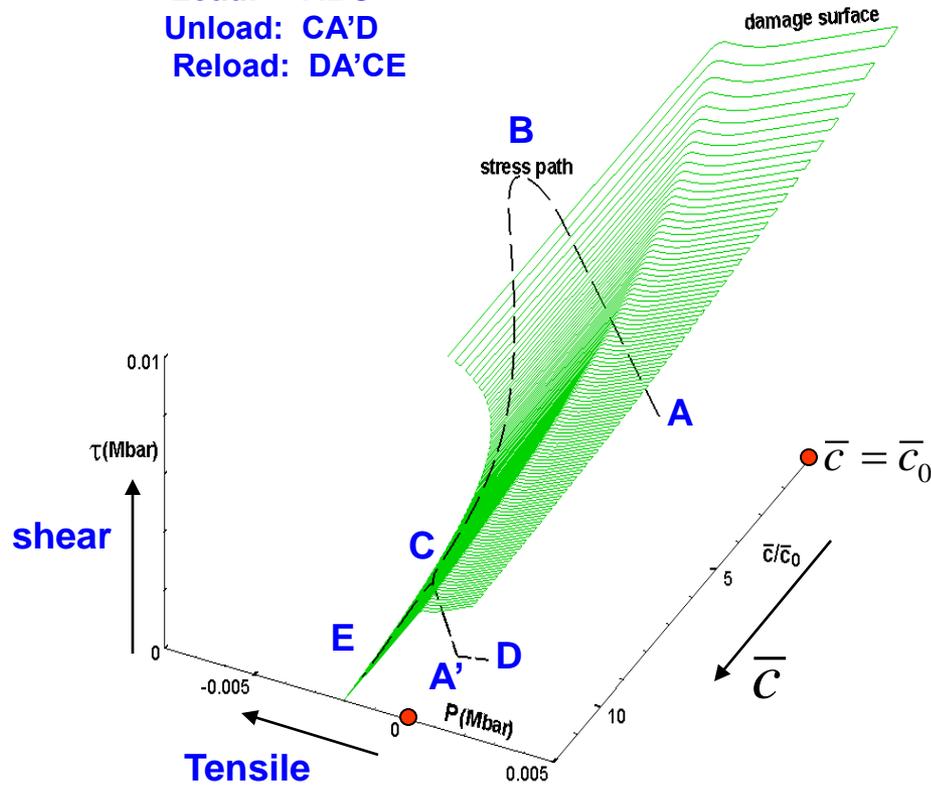
**Stress-strain**

**Compression is much stronger than tension**

$$\sigma_c / \sigma_t = 2\sqrt{1-\nu} / 2 \left( \sqrt{\mu^2 + 1} + \mu \right)$$

# Damage Surface and Stress path

Load: ABC  
Unload: CA'D  
Reload: DA'CE



The damage surface is continuous;

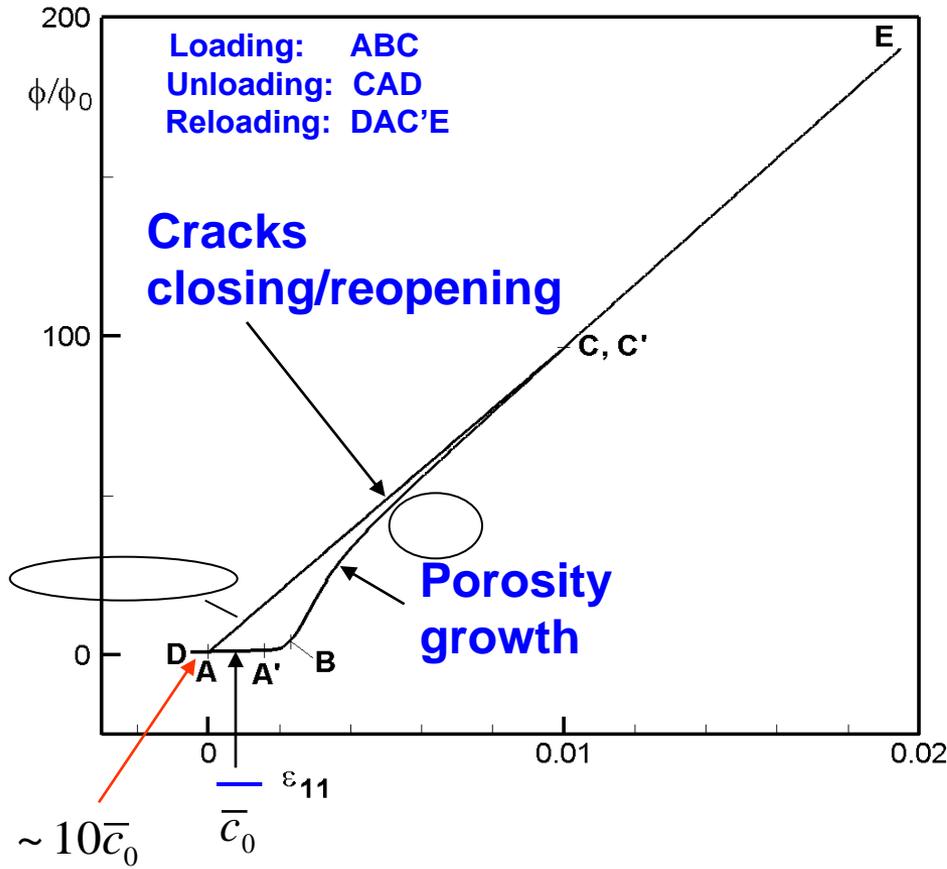
Size of the damage surface (in stress space) shrinks as cracks grow;

Stress path is above the damage surface, due to rate effects;

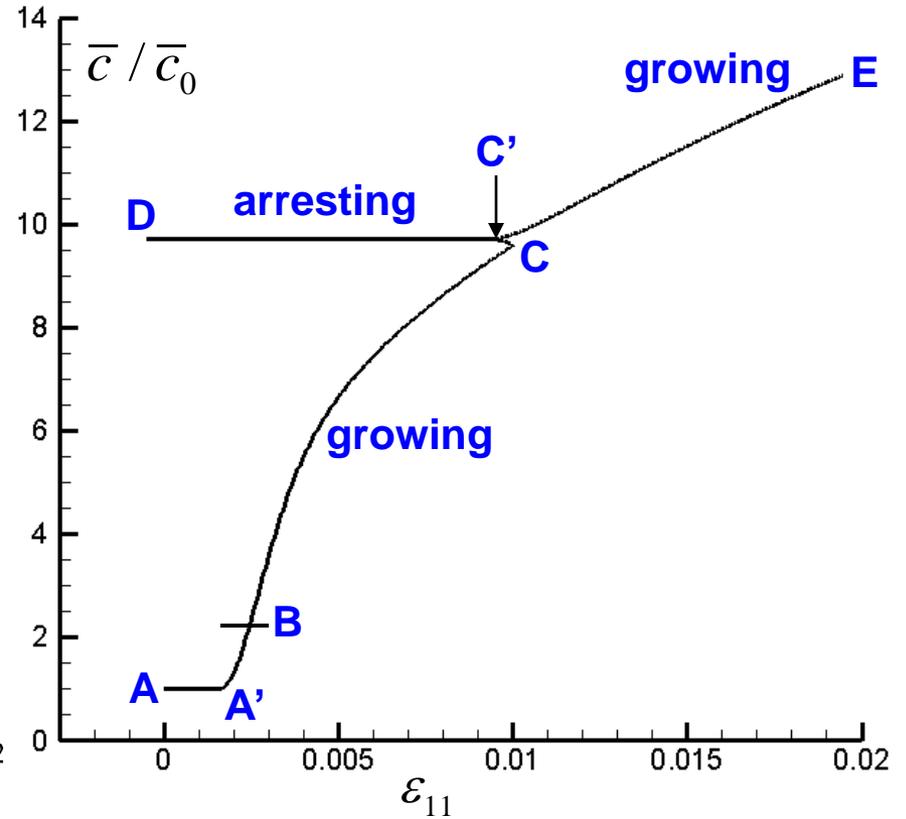
Pronounced strain softening right after the stress peak.

Uniaxial (strain) Tension loading

# Cyclic loading - II

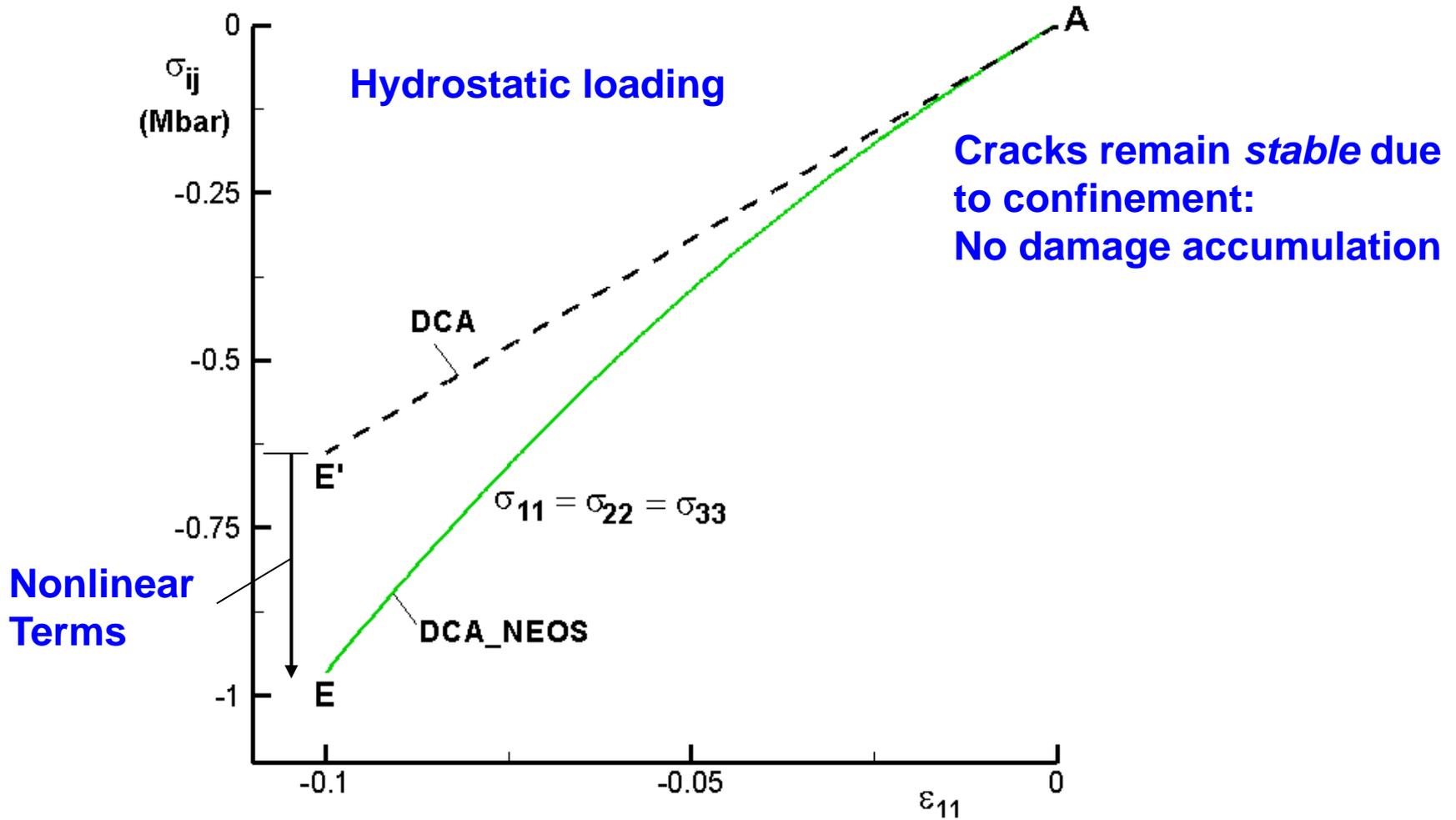


Evolution of porosity

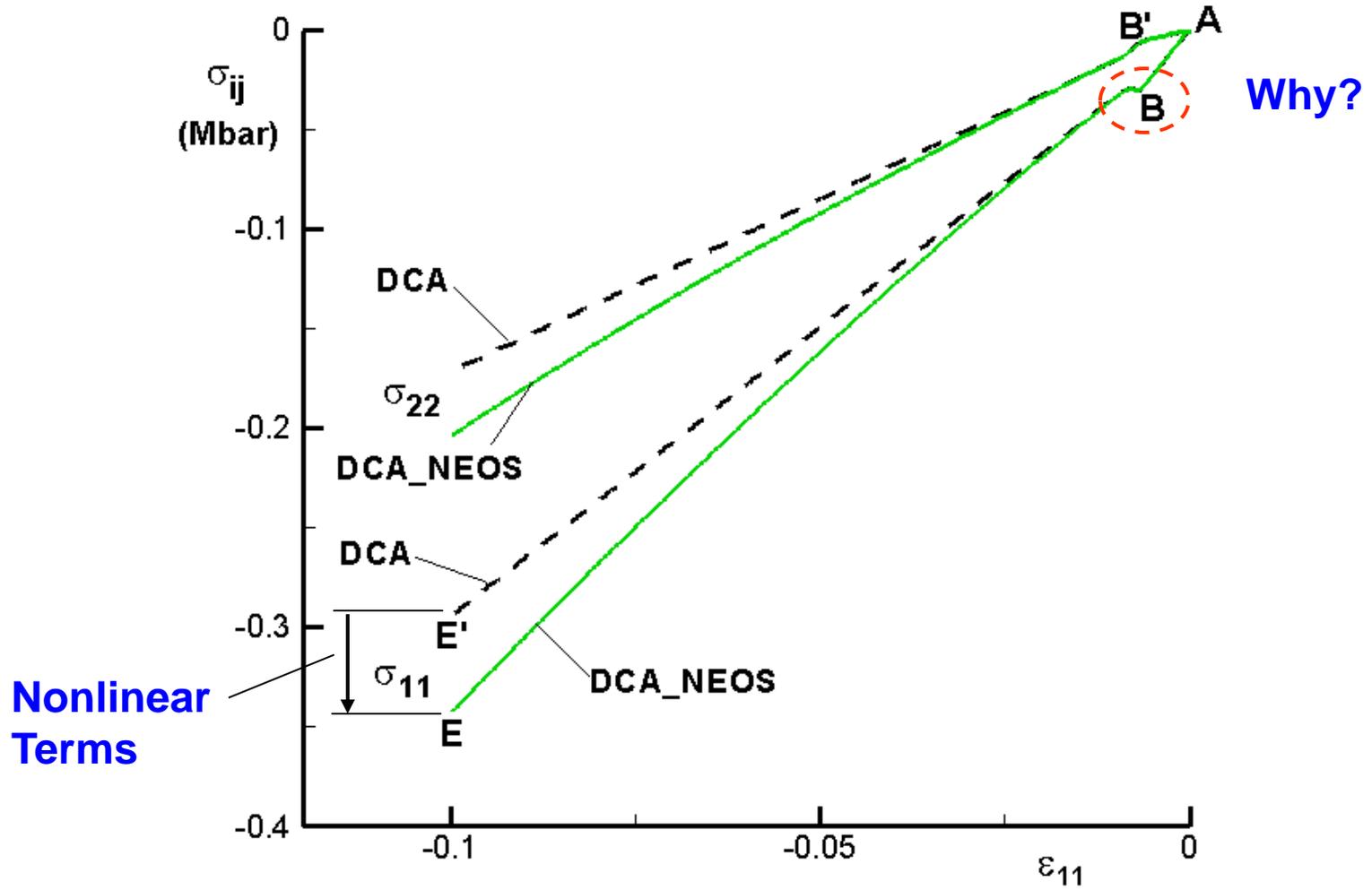


Evolution of crack size

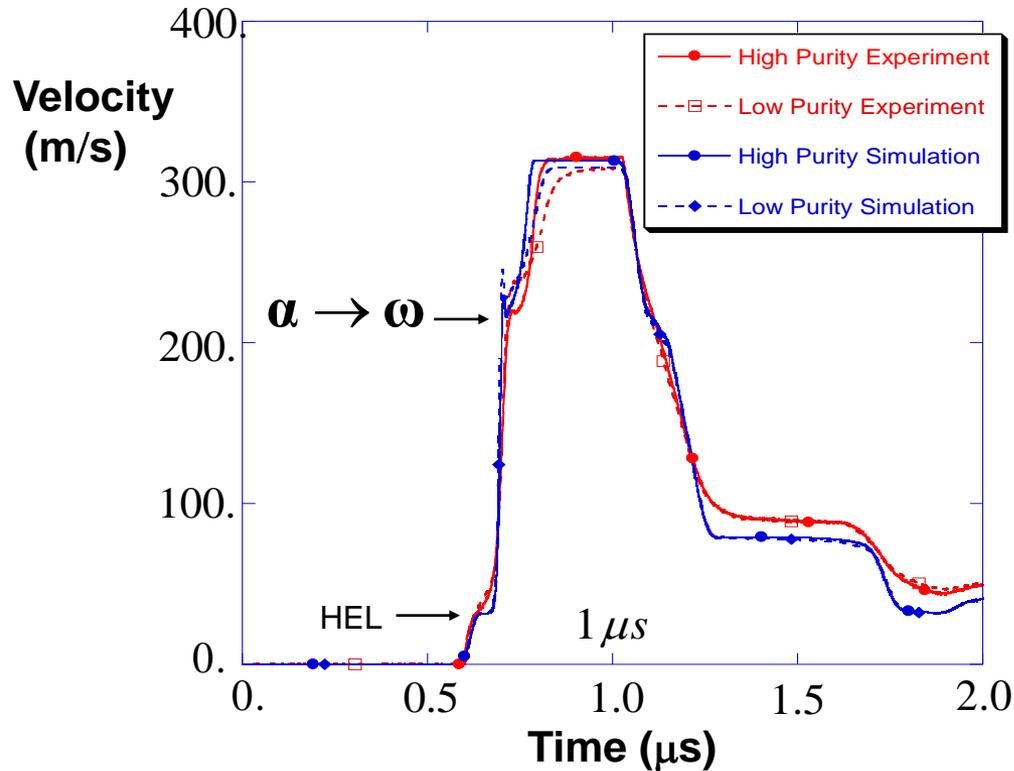
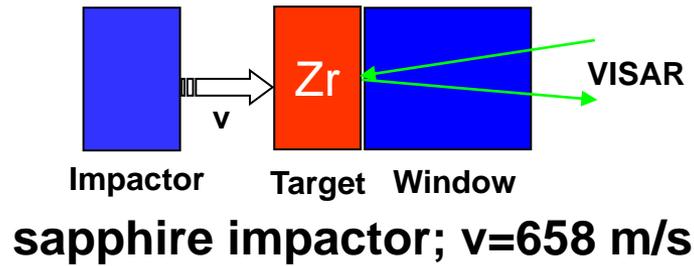
# Large strain (10%) compression: *Hydrostatic Loading*



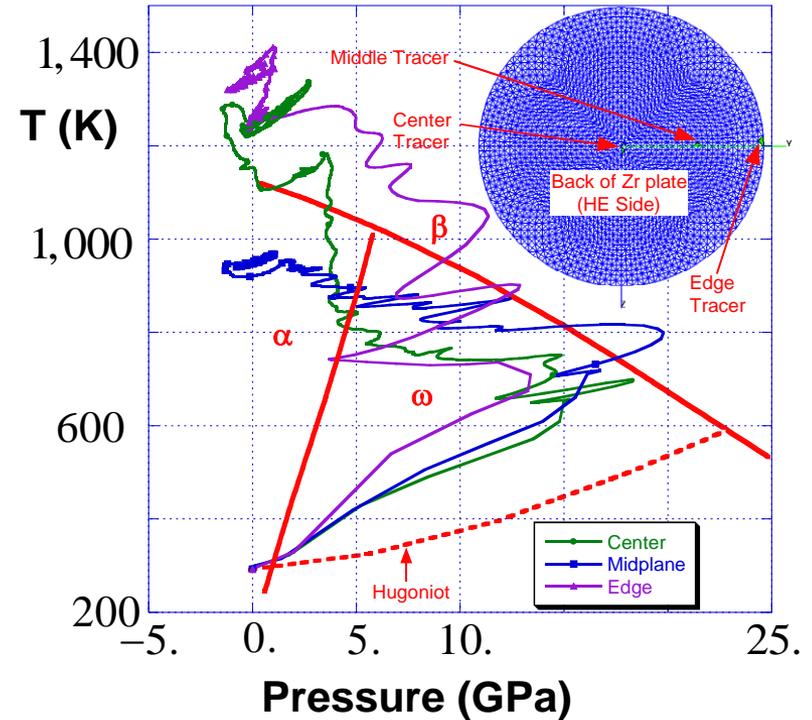
# Large strain compression: *Uniaxial strain*



# Multiphase Plasticity Model for Zirconium



## The phase diagram of Zr



That the material experiences all three phases can have important implications...